

# SUPPLEMENTAL TYPE CERTIFICATE 10051448, REV. 1

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

# FOKKER SERVICES B.V.

HOEKSTEEN 40 2132 MS HOOFDDORP NETHERLANDS

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: EASA.IM.A.191

Type Certificate Holder: BOMBARDIER INC

Type Design - Model: DHC-8-102/DHC-8-103/DHC-8-106

DHC-8-201/DHC-8-202

DHC-8-301/DHC-8-311/DHC-8-314

DHC-8-315

DHC-8-401/DHC-8-402

# Description of Design Change:

Upgrade of TCAS 7.0 to 7.1 on DHC-8 100/200/300/400 Series

#### **EASA Certification Basis:**

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

CS 25.1301(a)(b)(c)(d), CS 25.1309(a)(b)(c) (d),(g), CS 25.1581(a) - all at Amdt. 14

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval

See Continuation Sheet(s)

For the European Aviation Safety Agency,

Date of issue: 11 February 2015

Alain LEROY

**Head of Large Aeroplanes Department** 

Note:

The following numbers are listed on the certificate: EASA current Project Number: 0010032892-001

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## **Associated Technical Documentation:**

Installation must be in accordance with Fokker Engineering Bulletin EBDHC-8-0013.

Operation must be in accordance with AFM Supplement DHC-8-BTI-AFMS-S-001, Revision 2.

Continued Airworthiness unchanged in accordance with OEM ICA or later revisions of the above listed documents approved by EASA.

## **Limitations/Conditions:**

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

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Note:

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